

1 **Synthetic topography from the decameter to the centimeter scale on Mars for scientific**
2 **and rover operations of the ESA-Roscosmos ExoMars mission**

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23
24 **Keywords**

- 25 • Topographic modelling
- 26 • Martian surface
- 27 • Image simulation

34 **0. Abstract**

35 The ESA-Roscosmos ExoMars platform and rover mission will have complex interactions with
36 the martian surface. In order to plan and perform engineering and scientific operations, the
37 morphological characteristics of the terrain surrounding the rover need to be characterized from
38 the decameter to the centimeter spatial scale. The smallest possible features currently identifi-
39 able are typically >0.75 m in size, corresponding to three times the ground sampling resolution
40 of the High Resolution Imaging Science Experiment (HiRISE) camera on NASA's Mars Re-
41 connaissance Orbiter (MRO). We have developed a synthetic topography with a ground sam-
42 pling resolution of 0.01 m by integrating 1) a modeled topography of small-scale reliefs (*e.g.*,
43 rocks) into 2) a 0.25 m resolution digital elevation model (DTM) built by applying stereo and
44 shape-from-shading techniques to HiRISE data. The modeled topography of small-scale reliefs
45 is based on the extrapolation of the abundance and spatial distribution pattern of geological
46 features measurable in HiRISE data. Once we had completed the synthetic topography, we
47 simulated camera images of the landscape. The intention is that these views may enable a more
48 efficient planning of image acquisition and possibly a rapid interpretation of actual mission
49 images that would help the determination of geological processes not identifiable from orbit.
50 We determined that the cumulative fractional area covered by reliefs, *i.e.*, float blocks and
51 ridged outcrops, over the entire Oxia Planum landing site spans the range 0–30% with a mean
52 abundance of reliefs of $7 \pm 5\%$.

53

54 **1. Introduction**

55 One of the main objectives of the European Space Agency (ESA) and the Russian Roscosmos'
56 ExoMars 2022 is the search for physical and chemical biosignatures through the *in situ* study
57 of ancient, Noachian-aged (~ 4.1 Ga old), sedimentary material at the landing site, Oxia Planum
58 (Vago *et al.*, 2017). This mission is part of a program (Vago *et al.*, 2015) that also includes the
59 Trace Gas Orbiter (TGO), launched in 2016 (Svedhem *et al.*, 2018). ExoMars 2022 will deliver
60 a landing platform and a rover to the red planet, each hosting a suite of scientific instruments.
61 Organic compounds are likely to be protected from the damaging radiation environment pre-
62 vailing on the martian surface only if buried in the subsurface for long periods. The ExoMars
63 rover will characterize the subsurface with a sounding radar, a neutron detector, and a spec-
64 trometer hosted in the rover drill—the latter is capable of reaching 2 m depth (Ciarletti *et al.*,
65 2017; De Sanctis *et al.*, 2017; Mitrofanov *et al.*, 2017). Samples collected by the drill will be
66 analyzed by three successive analytical instruments to characterize their mineralogy, chemistry

67 and any organic compounds (Bibring *et al.*, 2017; Goesmann *et al.*, 2017; Rull *et al.*, 2017). A
68 suite of proximity (*i.e.*, surface remote sensing) instruments will investigate the geologic con-
69 text: a panoramic camera (PanCam), a close-up high-resolution imaging camera (CLUPI) and
70 a near infrared point spectrometer (Coates *et al.*, 2017; Josset *et al.*, 2017; Korablev *et al.*,
71 2017). PanCam consists of a set of stereo cameras named PanCam WAC (Wide Angle Cam-
72 eras) and a High Resolution Camera (HRC) (Coates *et al.*, 2017). Each PanCam WAC projects
73 a field of view (FOV) of $38^\circ \times 38^\circ$ into a 1024×1024 pixels wide detector. A filter wheel is used
74 to acquire multispectral images in the visible wavelength. PanCam HRC projects a FOV of
75 $4.8^\circ \times 4.8^\circ$ into a 1024×1024 pixels wide detector and makes use of a Bayer filter in front of the
76 sensor to enable color imaging. Camera used for engineering purposes are the NavCam, located
77 close to PanCam at the top of the mast, and LocCam located at a height of about 1 m at the
78 bottom of the mast and fixed at an angle of 18° downward (Silva *et al.*, 2013). NavCam and
79 LocCam are identical stereo cameras that use a detector 1024×1024 pixels in size with a FOV
80 of $68^\circ \times 68^\circ$. The CLUPI camera has a FOV of $12^\circ \times 8^\circ$ (14° diagonal) and is equipped with a
81 $2652 \times 1768 \times 3$ pixels sensor composed of 3 stacked layers of pixels (red, green, and blue) for
82 color acquisition (Josset *et al.*, 2017). CLUPI is mounted on the drill so that its height and angle
83 with respect to the horizon is variable. In addition, different FOV can be achieved with the use
84 of mirrors.

85 The ExoMars platform and rover will perform complex engineering and scientific op-
86 erations. The principal interactions with the geological environment required for engineering
87 purposes include: touchdown with the platform legs, imaging the landing area for determining
88 its geographic coordinates, deployment of two sets of rover ramps, rover egress, rover drive
89 (~ 60 meter) to exit the descent engine blast contamination zone, close approach of the rover to
90 an outcrop, surface/deep drilling and sample extraction, sample crushing, autonomous naviga-
91 tion for several tens of meter. For scientific purposes, knowledge of the surrounding geological
92 environment is a key requirement for science planning, *e.g.*, to understand the subsurface stra-
93 tigraphy and identify target depths where samples may be rich in organic compounds. The
94 subsurface stratigraphy will be elucidated by integrating and interpreting data from subsurface-
95 sensing instruments (the radar, neutron spectrometer, and drill-mounted infrared spectrometer)
96 together with observations of the surface geological environment (from PanCam, ISEM and
97 CLUPI). Performing combined science in this manner will maximize what can be learned about
98 the geologic history of drilled samples.

99 Knowledge of the geological environment needs to span a range of spatial scales, from
100 the tens of meters (*e.g.*, terrain slope during platform touchdown), to the centimeter scale (*e.g.*,
101 wheel/soil interactions) in order to plan and perform the variety of operations. A thorough
102 characterization of the local geology in its *sensu largo*, including, *e.g.*, material strength, grain
103 size distributions, composition, cannot be attained before the actual mission. However, we can
104 obtain a considerable amount of advanced information by concentrating on the topography of
105 the terrain. The topography datasets for NASA missions are constructed with stereo-photo-
106 grammetry using High Resolution Imaging Science Experiment (HiRISE) imaging (McEwen
107 et al., 2007). HiRISE Digital Terrain Models (DTMs) have a best ground sampling resolution
108 of ~ 1 m (Golombek et al., 2018). Ancillary information on rock abundance is calculated using
109 HiRISE imaging at the meter scale and a model dependent extrapolation to the centimeter scale
110 (Golombek and Rapp, 1997).

111 The missing coverage in spatial resolution, especially in the range critical for the safety
112 and traversability of lander and rover missions, *e.g.*, below HiRISE DTM resolution, is illus-
113 trated in Figure 1. Planetary surface topography represents the cumulative effect of geologic
114 processes operating over different spatial- and time-scales. Generally, slope distributions are
115 skewed towards steeper slopes at shorter baselines. Parameterizing the relationship between
116 slope distribution and baseline (*e.g.*, by fitting values of the Hurst exponent, Rosenberg et al.,
117 2011) permits the investigation of general topographic trends (Aharonson et al., 2001), repre-
118 sented by the power law scaling of slope abundance with baseline. A schematic of this rela-
119 tionship is illustrated in Figure 1. Also plotted in Figure 1 is the ground sampling resolution
120 that is possible using different Mars topographic datasets, and the types of geological features
121 responsible for the topographic expression. There is a spatial domain below ~ 1 m where the
122 slope distribution cannot be directly constrained with available data. Given the typical ground
123 clearance and mobility performance of Mars landers and rovers, it is this domain, between a
124 few cms and 1 m, that is critical to assess landing safety and terrain traversability.

125 This challenge has been considered in the work of Martin et al. (2014) for the case of
126 airless bodies, like the Moon, where the terrain morphology is dominated by impact craters. In
127 Martin et al. (2014), the regional topography below the resolution of the data-based DTM is
128 modeled with a purely theoretical fractal approach. The local topography due to impact craters
129 is obtained following two models: one for the shape of the initial crater and another for the
130 shape of the crater after diffusional degradation. Van Wal (2018) and Tardivel et al. (2014)

131 addressed a similar subject for small bodies with the frequency of blocks following a power
132 law distribution.

133 For Mars, studies have been performed recently to increase the spatial resolution and
134 reliability of DTMs. Among other studies (e.g., Gehrke, 2006, O’Hara and Barnes, 2012, Gupta
135 et al., 2014), the works of Jiang et al. (2017), Wohlfarth et al. (2018) and Hess et al. (2019a),
136 used the stereo and shape-from-shading techniques (e.g., Shao et al., 1991) to improve Martian
137 DTMs obtained with the stereo-photogrammetry method. Later, these methods have also been
138 applied to HiRISE imagery (e.g., Doute and Jiang, 2019; Hess et al., 2019b). The ground sam-
139 pling resolution of such high-resolution DTMs is ~ 0.25 m, and the smallest resolved features
140 are blocks of 0.7 m in width (Hess et al., 2019b).

141 In order to tackle the lack of topographic information at the centimeter level in the latter
142 DTM, we have developed a synthetic topography built by combining two datasets. The syn-
143 thetic topography is calculated by adding the topography of small-scale reliefs (e.g., meter and
144 sub-meter sized boulders) onto the high-resolution DTM. Importantly, the spatial pattern and
145 abundance of small-scale reliefs follows geological units (Figure 2). Because the geological
146 units are mapped by taking advantage of the highest resolution possible (i.e., segmented
147 HiRISE), the resulting topography is more representative of the actual terrain than has been the
148 case in previous approaches (e.g., Martin et al., 2014 for the Moon). In this article we will first
149 present the technique employed for calculating the high-resolution DTM, followed by the
150 method used in the construction of topography and pattern of the small-scale reliefs. We will
151 then produce an example of a synthetic topography and explain its validation and performance
152 (model vs. *in situ* rocks). Finally, cases using the synthetic topography and ancillary products,
153 e.g., rock abundance map, will be presented and discussed.

154

155 **2. Method**

156 **2.1 Used dataset**

157 Here we briefly present the technique that has been used to produce the high-resolution
158 DTMs. The reader is encouraged to consult Hess et al. (2019b), and references therein, for a
159 more thorough description. A common approach to improve the quality of planetary DTMs
160 relies on the refinement of an initial DTM with the means of Shape and Albedo from Shading
161 (SfS). Various methods have been devised with this principle such as Grumpe and Wöhler,
162 2014, Grumpe et al. 2014, Wu et al. 2017, Jiang et al. 2017, Alexandrov and Beyer 2018, Hess
163 et al. 2019b. The initial DTM can be either generated from laser altimetry data or more highly

164 resolved stereo-photogrammetric methods. The initial constraint ensures an accurate absolute
165 height while SfS can account for slopes and details on small scales. Using an initial stereo
166 DTM, SfS also allows to reduce the well-known artifacts produced by the block-matching
167 method and pixel locking (Gehrig and Franke, 2016) effect in stereo-photogrammetric prod-
168 ucts. In our case, this leads to an effective ground sampling resolution equal to the pixel scale
169 of the original image, *e.g.*, 0.25 meter (hereafter m) for HiRISE images.

170 The application of the shading procedure requires compensating for the effects of the
171 atmosphere. The atmospheric parameters of the combined reflectance and atmospheric model
172 are estimated by performing a nonlinear regression between the radiance data of the HiRISE
173 imagery and the initial stereo DTM. Based on an initial stereo DTM that is very accurate for
174 low and medium frequency components (*i.e.*, on a larger and medium spatial scales) the SfS
175 procedure iteratively refines the surface. At each resolution level, accurate modeling of the
176 reflectance behavior and the influence of the atmosphere, both mainly based on the Hapke
177 model (Hapke, 2002), is employed to change the orientation of the surface facets such that the
178 difference between measured and modeled reflectance values is minimized in the least square
179 sense (Hess et al., 2019a). Additionally, regularization terms are employed to solve the ill-
180 posed problem (Shao et al. 1991, Grumpe et al. 2014). Combining the accurate photogrammet-
181 ric reconstruction with the proper slope estimation that is performed by the SfS, the advantages
182 of both approaches can be harnessed. The accuracy of this procedure has been studied on the
183 MRO/Context Camera (CTX) resolution using HiRISE stereo images as ground truth (*e.g.*,
184 Hess et al. 2019a). The vertical accuracy achieved with this method was, in many cases, well
185 below 2 m at a pixel resolution of 6 m/pixel (hereafter m/px). If the implications of this vertical
186 accuracy vs. spatial resolution relationship are directly extrapolated to HiRISE images (capable
187 of a maximum resolution of 0.25 m/px), for which, unfortunately, no ground truth comparison
188 is possible before landing, the implied vertical accuracy can be estimated to be in the decimeter
189 range.

190

191 **2.2 Technique for the topography and spatial pattern of small-scale reliefs**

192 The reliefs detectable in HiRISE images are ~1 m to several m wide features that typi-
193 cally correspond to rootless blocks (*e.g.*, boulders) or to topographic ridges (*e.g.*, outcrops). In
194 other publications these features are referred to as ‘rocks’. However, because Oxia Planum has
195 a considerable fraction of small ridges in addition to boulders (Quantin-Nataf et al., 2020,
196 Ivanov et al., 2020), we use the more general term ‘reliefs’. Note that from an engineering point

197 of view, rootless boulders or ridges can be considered identical; *e.g.*, they can be equally dan-
 198 gerous at touchdown. The detection technique is based on shadows cast by features and closely
 199 follows the approach described in Golombek *et al.* (2003). The major difference in this study
 200 resides in the shadow segmentation method. The detection algorithm follows the following
 201 steps: (i) dataset preparation, (ii) calculation of derived products and shadow segmentation,
 202 (iii) estimation of the size of reliefs casting shadow at the m-scale, (iv) extrapolation of relief
 203 abundances at the sub-m scale. The technique has been developed based on visual investiga-
 204 tion of a few hundred reflectance profiles across shadows (Figure 3a) cast by a variety of land-
 205 forms (*e.g.*, blocks, ridges, crater rims, dunes, mesas) over a variety of terrains (*e.g.*, of varying
 206 albedo, aeolian cover, bedrock exposure) and with different ranges of digital numbers corre-
 207 sponding to different instrument conditions. The sun-synchronous orbit of MRO yields images
 208 with approximately constant illumination geometry.

209

210 **2.2.1 Data preparation**

211 HiRISE data products require pre-processing before applying the detection algorithm.
 212 HiRISE images from the Planetary Data System (PDS) are geographically registered and cali-
 213 brated to $I/F \cos(i)$, with i denoting the average incidence angle of the image. The image is
 214 divided into $250 \text{ m} \times 250 \text{ m}^2$ ($1000 \text{ px} \times 1000 \text{ px}$) sub-frames. Each sub-frame is rotated in or-
 215 der to have the sub-solar azimuth consistently at 270°W .

216

217 **2.2.2 Shadow segmentation**

218 The following procedures are applied to each sub-frame.

219 (1) A so-called “shadow depth” (SD) frame is calculated. This frame detects decreases in re-
 220 flectance independently of the local mean albedo. It exploits the property of a strong and spa-
 221 tially limited decrease in reflectance due to a small-scale shadow (Figure 3a). The calculation
 222 uses the parameter:

223

$$224 \text{ depth} = (R_{\text{continuum}} - R_{\text{shadow}}) / R_{\text{continuum}} \text{ (eq. 1)}$$

225

226 where R_{shadow} is the reflectance of the sub-frame pixel located at (x,y) . $R_{\text{continuum}}$ is a mean re-
 227 flectance outside the shadow calculated with vertical and horizontal pixels located at a distance
 228 d from the considered pixel (x,y) . Three shadow depth frames are calculated with $d = 5, 7, 9$
 229 pixels named $SD_{d=5}$, $SD_{d=7}$, $SD_{d=9}$, respectively.

230 (2) Each SD frame is segmented with the maximum entropy thresholding (*e.g.*, Kapur et al.,
231 1985). Note that Golombek *et al.* (2008) applied the same thresholding to reflectance sub-
232 frames and not to shadow depth frames. Each SD frame leads to a pixel mask that identifies
233 pixels in shadow.

234 (3) Pixels identified with shadow in each SD frame (*i.e.*, $SD_{d=5}$, $SD_{d=7}$, $SD_{d=9}$) are merged into
235 a single pixel mask. The use of three different SD masks allow to detect both small (~ 1 m size)
236 and large (~ 5 m size) shadowed regions.

237 (4) Isolated pixels with shadow are removed. Our visual investigation confirmed that single-
238 pixel detections are most frequently noise that does not represent shadow.

239 (5) The derived frame detects strong decreases in reflectance in the spatial range spanning from
240 1 m to several m. We found that such decreases could also be mimicked by albedo pattern,
241 usually dark dunes on bright bedrocks. Therefore, an additional frame is calculated to partially
242 avoid such pattern. By visually studying the shadows detected with shadow frames, we found
243 that the reflectance profile of shadow in the illumination direction is characterized by an abrupt
244 decrease in reflectance occurring within about three pixels (Figure 3a). Thus, we calculate a
245 frame ($SD_{d=1}$) with the same principle of the depth parameter, with, however, a distance $d=1$
246 and in the horizontal (illumination) direction only. The frame $SD_{d=1}$ is used in the next step.

247 (6) Contiguous pixels are identified in the mask. To partially avoid false positives due to albedo
248 pattern, only groups of pixels for which the $SD_{d=1}$ frame contains an abrupt decrease in reflec-
249 tance ($SD_{d=1} > 0.05$) are retained.

250

251 **2.2.3 Estimation of the size of reliefs**

252 The width of each remaining group is measured perpendicularly to the illumination
253 direction. This width is assumed to define the size of the relief. The shadow length measured
254 in the illumination direction is used to estimate the height of the feature, assuming it rests on a
255 flat surface. We found that for sizes 2.5 m and larger, depressions of impact craters and escarp-
256 ments start to dominate the detections. Disentangling blocks from other features is beyond the
257 capability of this method, therefore the decision was taken to disregard all detections having
258 size bigger than 2.5 m, as implemented in Golombek *et al.* (2012). As shown further below,
259 this cutoff does not affect the measurements at smaller sizes. At this stage, the size and location
260 of all small-scale topographic features across the frame are determined (Figure 4).

261

262 **2.2.4 Extrapolation of relief abundances at the sub-meter scale**

263 In order to estimate the abundance of features at the centimeter scale, we adopt the
264 technique described in and Golombek et al. (2003, 2008). The area where the extrapolation is
265 done can be either a sub-frame or a geological unit. Here we describe the technique using a
266 sub-frame. The use of a geological unit is described in section 2.3. Our approach is based on a
267 correlation between sub-meter abundances, as observed at the Viking lander sites, and the cor-
268 responding meter-scale HiRISE observations (Golombek et al., 2003). This correlation has
269 been confirmed to be valid also for other sites (Golombek et al., 2008), although not for all
270 (e.g., Heet et al., 2009). The variety of geological processing controlling reliefs formation,
271 coverage, and erosion is such that a single model is not sufficient to describe their abundance
272 across the entire martian surface. If developed, improved context-dependent models might be
273 used in the future. The fraction of surface covered by rocks with diameters equal to or greater
274 than some value D for various total rock coverage fractions is defined by:

275

$$276 \quad F(D) = k \exp[-q(k)D] \quad (\text{eq. 2})$$

$$277 \quad q(k) = (A + B/k) \quad (\text{eq. 3})$$

278

279 with $A=1.79$ and $B=0.152$ (Golombek and Rapp, 1997). The parameter k is the total rock cov-
280 erage or the cumulative fraction of surface covered by blocks of all sizes. The parameter $q(k)$
281 defines the rate of drop-off in the exponential function at large blocks diameter. We imple-
282 mented the technique with the following two steps.

283 First, for each sub-frame, the cumulative size frequency distribution of features per m^2
284 is calculated in the size range 1.5–2.5 m. The lower boundary is set by the spatial resolution
285 (Sefton-Nash et al., 2016, Golombek et al., 2003), whereas the upper boundary is defined in
286 section 2.2.3.

287 Second, a best fit between the measured cumulative size frequency distribution and the
288 model distributions of Golombek and Rapp (1997) is determined (Figure 3b) and results in the
289 estimation of the parameter k . Because in this work we include ridges in addition to rocks, we
290 refer to the parameter k^* , to distinguish it from the k calculated in previous studies for rocks
291 only.

292

293 **2.3 Technique for the development of synthetic topography**

294 The construction of synthetic topography necessitates the segmentation of the region of
295 interest into geological units. The unit boundaries are mapped visually using single HiRISE

296 images in the same manner as in geological mapping, with the ESRI ArcGIS software. A unit
297 in a geologic map is assumed to have been formed by a single geologic process or event (alt-
298 hough it may have been the subject of erosion and or other diagenetic effects thereafter). Thus,
299 the relief abundance it contains is most likely to be spatially homogeneous. Because these units
300 can be very small (area $\sim 100 \text{ m}^2$) the abundance of reliefs may prove too low to obtain a reliable
301 fit to the size frequency distribution to the Golombek and Rapp (1997) model. In such cases,
302 the abundance is evaluated visually, by comparing the unit with look up maps, *i.e.*, of larger
303 areas having known homogeneous abundance (Figure 5). Only with this approach was it pos-
304 sible to estimate the abundance of reliefs for extremely small areas ($\sim 100 \text{ m}^2$). This approach
305 is taken at the expense of the abundance precision. We estimate that the latter decreases relative
306 to what can be achieved in areas $\geq 60,000 \text{ m}^2$ (*e.g.*, sub-frame area) to about $\pm 5\%$.

307 The final step to produce synthetic topography consist in the addition of small-scale
308 reliefs to the stereo and shape-from-shading HiRISE DTM. In order to do this, we performed
309 a cubic interpolation to generate a DTM with a spatial resolution of 0.01 m based on the
310 HiRISE height information (0.25 m). Small scale reliefs for each unit were randomly distrib-
311 uted starting from larger to smaller diameters according to the size frequency distribution cal-
312 culated in previous steps. In order to avoid spatial overlaps, reliefs were iteratively randomly
313 redistributed if they would superpose existing reliefs or exceed unit boundaries. We considered
314 simple shapes for the reliefs, *i.e.*, square in planar view and aspect ratio of 0.5 (height/width),
315 as well as a dome shaped structure.

316

317 **3. Results**

318 We obtained small relief abundance maps across Oxia Planum where sufficient data was avail-
319 able (Figure 6). Manual geological mapping to produce the final synthetic topography is a time-
320 consuming effort. Thus, we limited our analysis to an area close to the center of the ExoMars
321 2020 landing ellipses with sufficient topography for our display purposes (Figure 7). The
322 touchdown target for the ExoMars 2022 landing site remains the same as for a launch in 2020,
323 only the azimuth of the dispersion ellipses are shifted. Because a large portion of the landing
324 ellipse is relatively flat, the synthetic topography was calculated in an area chosen to highlight
325 the effectiveness in reconstructing topographic reliefs. This area is not representative of the
326 most likely landing terrain. It is instead characterized by slopes $\leq 8^\circ$ (over a few m) and relief
327 abundances up to a cumulative fractional area of 20%.

328

329 4. Discussion

330 Here we first discuss the algorithm for rock detection, followed by a discussion of the recon-
331 structed synthetic topography.

332 The precision of the automatic identification of blocks was tested by estimating the
333 cumulative fractional area at the landing sites of the Viking Landers 1 and 2, Mars Pathfinder,
334 Phoenix, Mars Science Laboratory (Gale Crater), Mars Science Laboratory candidate site
335 (Eberswalde), and Insight. There is an agreement within 5 % between our algorithm and liter-
336 ature estimates (Golombek and Rapp, 1997; Golombek et al., 2008, 2012) (Figure 8). This is
337 the largest (*i.e.*, worst case) difference because the comparison could not be made with exactly
338 the same dataset. For some counts presented in the literature, the precise location of the count
339 area was not indicated, the size of the area is different, or the image (HiRISE observation num-
340 ber) is unknown and might be different. Other discrepancies must arise from the image en-
341 hancement and the detection algorithm itself. For example, we use a best-fit approach to esti-
342 mate the cumulative fractional area, whereas the Golombek et al. (2008) method employs
343 lookup tables. In order to test our algorithm with actual rock abundances, we compare in Ta-
344 ble 1 the cumulative fractional area of rocks 1) from lander observations, 2) from our algo-
345 rithm, and 3) from the canonical Golombek et al. (2008) method. The landing sites having
346 sufficient rock abundance for which it is possible to perform estimates with HiRISE images
347 are few; they are limited to the Viking landers and Mars Pathfinder locations. As in the previous
348 comparison, the size of the count areas is different between the lander and the orbital observa-
349 tions.

350 We have found that at Oxia Planum very strong differences in reflectance (~ 0.1 in lam-
351 bertian albedo) can occur within very small areas ($\sim 5 \text{ m}^2$). These variations in reflectance are
352 equal or larger than that between shadowed and non-shadowed surfaces. These very small areas
353 consist of aeolian deposits (*e.g.*, longitudinal dunes) of dark material on top of high reflectance
354 and dust-free bedrock. In some instances, the dark aeolian material can be found filling narrow
355 fractures of the bedrock. The stark and spatially abrupt reflectance difference found in these
356 small areas leads to artifacts. We have found that both the algorithm for the high resolution
357 DTM and for the shadow detection are affected by this type of feature. The extreme similarity
358 between these features and shadow casting reliefs, and their relative rarity within Oxia Planum,
359 led us not to consider an *ad hoc* workaround to avoid them.

360 Another potential issue has its origin in HiRISE images. Although HiRISE images may
361 appear without artifacts in a visual investigation, and are thus usable for photo-geological anal-
362 ysis, they might have distinct contrast that lead to differences in shadow detection. This is

363 shown in some HiRISE images in Figure 6. The reason for such differences might have an
364 origin in the instrument level (gain), or be related to atmospheric conditions, although clouds
365 or dust coverage were not visually identifiable. Differences in rock abundance in juxtaposing
366 HiRISE images is also seen in other studies (Figure 9a in Golombek et al., 2012).

367 A comparison of the synthetic topography with actual data can be done with a single
368 dataset, *i.e.*, images acquired by the Mars Descent Imager (MARDI) onboard the NASA Curi-
369 osity rover (Malin et al., 2017). The camera acquired optical images of the surface during the
370 landing phase with a ground sampling resolution at and below the centimeter scale. The image
371 in Figure 9a has a ground sampling resolution of 0.1 m/px, in Figure 9c of ~ 0.05 m/px. Units
372 common to both images are topographically flat, likely representing aeolian deposits, and
373 rough units composed of ridges with varying amounts of blocks. We note how in both synthetic
374 and actual images the flat units are free of blocks. Shallow depressions as small as 5 m in width
375 are present at Gale crater and correspond to subdued impact craters. The synthetic image pre-
376 sent shallow (<10 cm deep, ~ 5 m wide) irregular depressions, but do not have the circular form
377 typical of impact craters. Such small impact craters might not be fully resolved, or they might
378 not be present at the location of the synthetic image. Using HiRISE images, Golombek *et al.*
379 (2012) estimated a cumulative fractional area (CFA) covered by blocks $< 10\%$ at the Curiosity
380 landing site. By qualitatively comparing the abundance of rocks in the MARDI image (Figure
381 9b,d) with the synthetic patterns, we find that the CFA in the 100×100 m² area centered at the
382 Curiosity touchdown location varies between 5% and 2.5%.

383

384 **5. Applications**

385 **5.1 Landing site safety**

386 The mean cumulative fractional area occupied by shadow-casting obstacles at the Oxia Planum
387 landing site is $7 \pm 5\%$. This estimate agrees with manual counts performed in discrete locations
388 as part of the landing site qualification tasks (*e.g.*, Sefton-Nash et al., 2016). On a qualitative
389 basis, there is agreement between the spatial distribution of blocks mapped manually (Pajola
390 et al., 2017) and with our automatic procedure. Oxia Planum has two major geological units: a
391 relatively smooth unit with a near infrared signature indicating the presence of Fe/Mg phyllo-
392 silicates, and an overlaying, dark, relatively rougher (at the tens of m scale) unit without clay
393 spectral components (Quantin-Nataf et al., 2020, Ivanov et al., 2020). In terms of abundance
394 of small-scale reliefs, we find that both units have a considerable range of variation and that
395 the clay-rich unit has only slightly lower relief abundances (Figure 10). Thus, the bulk area of

396 the two units can be considered similar from an engineering point of view (*e.g.*, landing safety
397 and traversability). On the other hand, the wind-eroded edges of the clay-free unit terraces
398 tend to be cliff-forming.

399

400 **5.2 Simulated camera images**

401 Model images of the synthetic terrain, as if taken by the ExoMars rover cameras, namely
402 NavCam, LocCam, PanCam and PanCam HRC (high-resolution camera), are shown in Figure
403 11. Note that these images are complementary to previous PanCam image simulations that used
404 a three-dimensional rendering system for the ExoMars rover without any information on its
405 surroundings (Miles et al., 2020). Figure 12 shows similarly modelled CLUPI images. The
406 scene depicted in our simulations represents a site of bedrock exposure due to aeolian erosion,
407 with consequent rock fragmentation, and localized aeolian deposition of sand (*e.g.*, Ivanov et
408 al., 2020). These simulated scenes are based on a geological context found in previous, alt-
409 hough not in all, Martian missions. They provide a starting point from which geological anal-
410 ysis of future actual images can be developed. Any difference with actual images would signal
411 the action of additional geological processes that could not be hypothesized based on the in-
412 formation derived from orbital investigations (Stack et al., 2016). For example, if the modeled
413 image corresponds to a landscape dominated by surficial basaltic deposits reworked by impact
414 and eolian processes (as observed, for example, by the rover Spirit at Gusev Crater, Grant et
415 al., 2004), then the artificial topography will be mostly characterized by blocks with an aspect
416 ratio of ~ 0.7 (b/a) and ~ 0.5 (c/a), where $a > b > c$ are the axis in order of decreasing length (*e.g.*,
417 Michikami et al., 2016). If, instead, in the mission image we observe regions with stacked rocks
418 having a very low height/width ratio, these might hint to an outcrop recording a different geo-
419 logical process (*e.g.*, fluvial sediments like those observed by Curiosity in Yellowknife bay,
420 Grotzinger et al., 2014). Such an outcrop, comprised of rocks that formed in the presence of
421 liquid water, would represent a region of prime science interest for a mission like ExoMars.
422 Obviously, this hypothetical preliminary insight from rock morphology only will need to be
423 confirmed by observations of rock textures and other properties. The comparison between the
424 actual and modelled images during a mission, where a visual discrepancy between the two
425 images can quickly signal the presence of an outcrop comprised of fluvial sediment could be
426 helpful to recognize regions of interest where to drive the rover. We clarify that this approach
427 is not proposed in substitution of – or for automatizing – a traditional visual investigation of

428 the acquired images, but only as an additional tool to increase the chances of not missing in-
429 teresting outcrops. Compositionally different lithologies are likely to occur in the same, mor-
430 phologically homogeneous, geological landscape and would not be detected with the approach
431 described above.

432 Moreover, the modelled PanCam HRC and CLUPI images may prove useful for interpreting
433 distant features that are poorly discerned in rover mission images (*e.g.*, far away objects or
434 blocks that are partially obscured by others). Indeed, the three-dimensionality of distant fea-
435 tures is sometimes difficult to assess. For example, from a single image taken under specific
436 light conditions, it might be difficult to conclude whether two visually discrete features are
437 actually boulders with dissimilar compositions or the perceived variation is only an artifact
438 created by different illumination of the same rock formation. A modeled image of the same
439 field of view might be helpful for discriminating between the two possible interpretations.
440 Figures 11 and 12 allow to better understand and appreciate the imaging capabilities of the
441 cameras regarding features likely to be present at the site. Because the cameras are pointing in
442 the same direction, we find that the set of cameras have complementarity and partial
443 redundancy (NavCam with PanCam, PanCam HRC with CLUPI) with respect to geological
444 targets. We note that the color capabilities of HRC and CLUPI are not taken into account here.

445

446 **5.3 Traversability planning**

447 We developed a simple traversability map based on the approximation that rover motion pro-
448 gress vs. time is a function of the local slope and the height of the small-scale reliefs it traverses.
449 Rover speed decreases with increasing slope abundance according to an exponential function
450 defined such that speed tends toward zero for a local slope of 30° or for a relief height of 30
451 cm. In this way, we computed a transformation from the synthetic topography into a traversa-
452 bility map. We considered the example case in which two potential science targets have been
453 identified on the slope of a small hill. The rover speed on a horizontal surface was set at 4 cm/s
454 (Zhou et al., 2013). Given the same rover starting point (at the beginning of sol), the objective
455 of the test was to calculate the travel time to the two locations. We found that the paths, opti-
456 mized for travel time, to the two targets have different lengths (95 m, 113 m) and can be nego-
457 tiated in approximately the same time (6h0min vs. 6h8min, respectively) due to roving across
458 different types of units (Figure 13).

459

460 **6. Conclusions**

461 The scientific and engineering work to be performed with the ExoMars Rosalind Franklin rover
462 will require knowledge of the topography at a spatial resolution that cannot be reached using
463 cameras on orbiting spacecraft. The planning of activities, such as landing site characterization
464 and testing of rover mobility on a representative terrain, is limited by the lack of a topographic
465 map at the centimeter scale. We have shown that a geologically meaningful extrapolation of
466 topography at relevant spatial resolutions can be achieved by combining high-resolution
467 HiRISE DTM, HiRISE segmentation based on geologic units, and a model of small-scale relief
468 abundances.

469 To calculate the synthetic topography for this work, we performed the detection of
470 small-scale reliefs on the Oxia Planum landing site. Our work indicates that the cumulative
471 fractional area covered by small-scale features is $7\pm 5\%$, consistent with manual counts carried
472 out at discrete locations.

473 In the near future, we expect to improve the visual realism of our simulated rover cam-
474 era images by considering more natural shapes for the small-scale reliefs based on the topog-
475 raphy of outcrops found on terrestrial analog sites.

476 Synthetic topography can be used also for missions involving long traverses, such as
477 NASA's Mars 2020 Perseverance rover (*e.g.*, Williams et al., 2020). The helicopter on Mars
478 2020 (Golombek et al., 2020) will capture images at the centimeter resolution. This will pro-
479 vide an ideal dataset to further test and improve this technique.

480

481

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484 Space and Technology Center (ESTEC) and by a Sofja Kovalevskaja Award of the Alexander
485 von Humboldt Foundation.

486

487 **8. Data availability**

488 The synthetic topography data is freely available at the following address or upon request at
489 Ottaviano Ruesch: <https://uni-muenster.sciebo.de/s/N6yEw5nM20Yd4H9>.

490

491 **9. Table and figures captions**

492

493 Table 1. Cumulative fractional area covered by rocks (in %) at three landing sites. Estimates
494 from ground truth observations and two HiRISE-based extrapolation methods.

495

496 Figure 1. Schematic plot showing the applicability of factors and DTM types at different spatial
497 scales (top), and the approximate placement on slope and size axes of common Mars geologic
498 features (bottom). The need to fill the data gap at ~sub-meter scales, where features are still
499 large enough to be relevant to landing and traversability, is highlighted. This domain is where
500 synthetic topography can play a role.

501

502 Figure 2. Workflow for the calculation of a synthetic topography using inputs from HiRISE
503 images, a stereo and shape from shading technique, and a model for the abundance of small-
504 scale reliefs.

505

506 Figure 3. (a) Reflectance profile across a block and its shadow from HiRISE data. Illumination
507 direction from the left. (b) Cumulative fractional area covered by small reliefs larger than a
508 given diameter, as a function of the relief diameter. Curves represent the distribution of reliefs
509 at 5, 10, 20, 30, 40 % from Golombek and Rapp (1997). Data points are from a 500 by 500 m²
510 area. Downturn at low diameter is a resolution effect.

511

512 Figure 4. (a) Shadow mask of a 500 by 500 m² area. (b) Contours (in red) of identified shadow
513 belonging to blocks overlaid on an HiRISE frame.

514

515 Figure 5. Representative areas for different abundances of shadow-casting reliefs expressed as
516 k^* (see text for definition). Abundance maps are color-coded (See figure 5 for labeling). Abun-
517 dance maps are 500 m wide. HiRISE close-ups are 110 m wide.

518

519 Figure 6. Color-coded maps of cumulative fractional area of reliefs superimposed on a High
520 Resolution Stereo Camera (HRSC) mosaic (Gwinner et al., 2016, 2019). The landing site el-
521 lipses for a 2020 launch are shown for different landing probability (light gray: 3-sigma, me-
522 dium gray: 2-sigma, dark gray: 1-sigma). The azimuth of the ellipses is a function of the launch
523 date.

524 Figure 7. Perspective view of a shaded relief from the synthetic topography at a spatial resolu-
525 tion of 1 cm over an area of 100 x 100 m². Red contour lines are shown every meter.

526

527 Figure 8. Comparison between cumulative fractional area covered by reliefs (k^*) at different
528 landing sites on Mars, measured in this study and in literature works. The values used in the
529 comparison are taken from plots and maps in the literature and are shown here with an uncer-
530 tainty of $\pm 2.5\%$.

531

532 Figure 9. Qualitative comparison of the spatial pattern of roughness units between the synthetic
533 topography at Oxia Planum shown as shaded relief (a,c) and a Mars Descent Imager (MARDI)
534 image acquired during descent of the Curiosity rover at Gale crater (b,d). Common units are
535 topographically flat and blocks-free ('F') and rough with varying amounts of blocks ('B').
536 Depressions are labeled 'D'. The synthetic image has units of different cumulative fractional
537 area covered by blocks (CFA). The CFA at the Curiosity landing site has been estimated with
538 HiRISE images to less than 10 % (Golombek et al., 2012). See main text for discussion.

539

540 Figure 10. Distribution of the abundance of small-scale reliefs (k^*) for the two major geological
541 units at the Oxia Planum landing site: a clay-rich unit (blue) and a dark unit (red).

542

543 Figure 11. Modelled images by the ExoMars rover cameras. (a) NavCam, FOV $68^\circ \times 68^\circ$. (b)
544 LocCam, FOV $68^\circ \times 68^\circ$. (c) One of two stereo images acquired by PanCam WAC, FOV
545 $38^\circ \times 38^\circ$. (d) PanCam HRC, FOV $4.8^\circ \times 4.8^\circ$. All images are 1024x1024 pixels in size. Percent-
546 age values indicate fractional area covered by rocks. Dashed lines indicate boundary between
547 different rock densities. The facet of a rock identifiable in all images is shown with a red box.
548 The synthetic topography is illuminated with a light source at an angle of 60° above the hori-
549 zon. Sky shown in white.

550

551 Figure 12. (a) CLUPI image in the geological environment survey mode with stowed drill and
552 use of the primary mirror. The height of CLUPI is 40 cm. The entire FOV of $12^\circ \times 8^\circ$ is pointing
553 20° downwards from a horizontal plane and ahead of the rover. The synthetic topography is
554 illuminated with a light source at an angle of $<10^\circ$ above the horizon to highlight cm-sized
555 reliefs. Image 2652x1768 pixels in size. (b) Diagram of the position of CLUPI and the drill
556 (yellow) corresponding to image acquisition in (a). (c) CLUPI image taken after the drill is
557 lifted and rotated, and is pointing horizontally from a 1 m height. Image 2652x1128 pixels in
558 size restricted to a FOV of $12^\circ \times 5^\circ$ using the second mirror. Here the synthetic topography is

559 illuminated with a light source at an angle of 60 ° above the horizon. (d) Diagram showing the
560 lifted and rotated drill enabling CLUPI images at various elevations. Computer-aided design
561 rendering (b) and (d) by Space Exploration Institute.

562

563 Figure 13. Rover traversability analysis on a color-coded synthetic topography. The triangle
564 symbol is the start point and stars are target locations. Green lines are paths with the shortest
565 traverse time calculated by considering the slope of the terrain and obstacles by small-scale
566 reliefs.

567

568

569 **10. References**

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